# M-G365PDC1/PDF1



## **IMU** (Inertial Measurement Unit)

## **■** GENERAL DESCRIPTION

The M-G365PDC1/PDF1 is a small form factor inertial measurement unit (IMU) with 6 degrees of freedom: triaxial angular rates and linear accelerations, and provides high-stability and high-precision measurement capabilities with the use of highprecision compensation technology. A variety of calibration parameters are stored in memory of the IMU, and are automatically reflected in the measurement data being sent to the application after the power of the IMU is turned on. With general-purpose SPI/UART support for host communications, the M-G365PDC1/PDF1 reduces technical barriers for users to introduce inertial measurement and minimizes design resources to implement inertial movement analysis and control applications. The features of the IMU such as high stability, high precision, and small size make it easy to create and differentiate applications in various fields of industrial systems.

### **■ FEATURES**

Small Size, Lightweight : 24x24x10mm, 10grams

Low-Noise, High-Stability

Bias Instability : 1.2 °/h Angular Random Walk : 0.08 °/√h

Initial Bias Error :  $360 \, ^{\circ}/h \, (1\sigma) \, / \, 3mG \, (1\sigma)$ 

6 Degrees Of Freedom

Triple Gyroscopes : ±450 °/s,

Tri-Axis Accelerometer : ±4 G (PDC0) / ±10 G (PDF0)

16/32bit Data Resolution

Digital Serial Interface : SPI / UART Calibrated Stability (Bias, Scale Factor, Axial Alignment) Data Output Rate : to 2k Sps

External Trigger Input / External Counter Reset Input

Delta Angle/Delta Velocity Output

**Attitude Output Accuracy** : ±0.2°

Calibration Temperature Range : -40°C to +85°C Operating Temperature Range : -40°C to +85°C

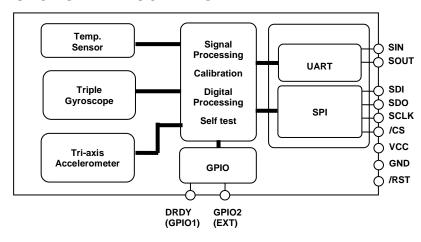
Single Voltage Supply : 3.3 V

Low Power Consumption : 16mA (Typ.)

## **■** APPLICATIONS

- Antenna Platform Stabilization
- Camera Gimbals
- **Navigation Systems**
- Vibration Control and Stabilization
- Pointing and Tracking Systems
- Autonomous Vehicle

## ■ FUNCTIONAL BLOCK DIAGRAM





## **■ SENSOR SECTION SPECIFICATION**

T<sub>A</sub>=25°C, VCC=3.3V, angular rate=0 °/s, ≤±1G, unless otherwise noted.

Parameter	r rate=0 °/s, ≤±1G, unless oth  Test Conditions / Comments	Min	Typ	Max	Unit					
GYRO SENSOR	Trest Conditions / Comments	IVIIII	ן יאָף ן	Ινιαλ	Offic					
Sensitivity										
Output Range	T		±450		°/s					
Scale Factor	16bit	-0.2%	66	+0.2%						
Ocale i actor	32bit	-0.2%	66x(2^16)	+0.2%	LSB/(°/s)					
Nonlinearity1 *7	1 σ, <300 °/s	- 0.270	0.05	- 10.270	% of FS					
(Best fit straight line)	1 σ, >300 °/s		0.2	_	% of FS					
Misalignment	1 $\sigma$ , Axis-to-axis, $\Delta$ = 90° ideal		0.01		0					
Bias										
Initial Error	1 σ, −40°C ≤ T <sub>A</sub> ≤ +85°C	T	360	_	°/h					
Repeatability	1 σ, turn-on to turn-on *3	_	36		°/h					
Bias Instability	Average	_	1.2		°/h					
Angular Random Walk	Average	_	0.08	_	°/√h					
Linear Acceleration Effect	Average	_	18	_	(°/h)/G					
Noise Density	f = 10 to 20 Hz	_	6.9	_	(°/h)/√Hz, rms					
Frequency Property	· ·									
3 dB Bandwidth		_	472	_	Hz					
ACCELEROMETERS										
Sensitivity										
Output Range	PDC1	_	±4	_	G					
	PDF1	_	±10	_	G					
Scale Factor	PDC1: 16bit	-0.1%	6.25	+0.1%						
	PDF1: 16bit	-0.1%	2.5	+0.1%	LSB/mG					
	PDC1: 32bit	-0.1%	6.25x(2^16)	+0.1%	LOD/IIIO					
	PDF1: 32bit	-0.1%	2.5x(2^16)	+0.1%						
Nonlinearity	PDC1 : 1 σ, <2G	_	0.1	_	% of FS					
(Best fit straight line)	PDF1 : 1 σ, <5G				0					
Misalignment	1 σ, Axis-to-axis, Δ = 90° ideal	_	0.01	_						
Bias		I	1 0 1							
Initial Error	1 $\sigma$ , $-40^{\circ}$ C $\leq$ T <sub>A</sub> $\leq$ +85 $^{\circ}$ C	_	3	_	mG					
Repeatability	1 σ, turn-on to turn-on *3	_	3		mG					
Bias Instability	PDC1 : Average	_	14	_	μG					
Velocity Random Walk	PDF1 : Average	_	16	_						
	PDC1 : Average	_	0.02 0.033	_	(m/s)/√h					
-	PDF1 : Average PDC1 : f = 10 to 20 Hz		48							
Noise Density	PDF1 : f = 10 to 20 Hz		80	<u> </u>	μG/√Hz, rms					
Frequency Property	DI 1 . I = 10 to 20 IIZ		1 00 1							
3 dB Bandwidth		T	167	_	Hz					
ATTITUDE OUTPUT		1								
Dynamic Range	Inclination Mode	-80		+80						
	Euler Mode ANG1:Roll	-45		+45	-					
	ANG2:Pitch	-180	_	+180						
	ANG3:Yaw*4	-180	_	+180						
Scale Factor		_	0.00012207	_	rad/LSB					
	16bit	_	0.00699411	_	°/LSB					
A · · · · · · · · · · · · · · · · ·	1 σ, Static	_	0.2		o					
Accuracy*4*6	1 σ, Dynamic *5 (100 °/s, max)		0.2							
TEMPERATURE SENSOR										
Scale Factor *1*2	Output = 2634(0x0A4A)		-0.0037918	<u> </u>	°C/LSB					
					1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1					

- \*1) This is a reference value used for internal temperature compensation. There is no guarantee that the value gives an absolute value of the internal temperature.
- \*2) This is the temperature scale factor for the upper 16bit (TEMP\_HIGH).
- \*3) Turn-on to turn-on / Day by day, estimated variation during 5 consecutive days.
  \*4) Yaw axis is not compensated for errors caused by drift.
- \*5) Dynamic accuracy is based on measurement data that has been measured from a stationary state. \* The accuracy that can be achieved depends on the input movement.
- \*6) Attitude output accuracy is based on measurement data for GLOB\_CMD2[0x16(W1)],bit[5:4]= 00: modeA.

Note) The values in the specifications are based on the data calibrated at the factory. The values may change according to the way the product is used.

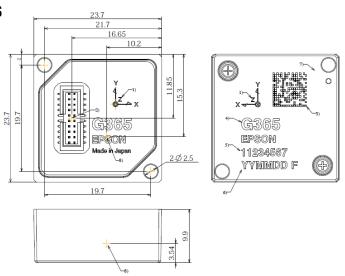
Note) The Typ values in the specifications are average values or  $1\sigma$  values.

Note) Unless otherwise noted, the Max / Min values in the specifications are design values or Max / Min values at the factory tests.

## ■ RECOMMENDED OPERATING CONDITION

Parameter	Condition	Min	Тур	Max	Unit
VCC to GND		3.15	3.3	3.45	V
Digital Input Voltage to GND		GND	_	VCC	V
Digital Output Voltage to GND		-0.3	_	VCC	V
				+0.3	
Calibration Temperature Range	Performance parameters are applicable	-40	_	85	°C
Operating Temperature Range		-40	_	85	°C

## **■ OUTLINE DIMENSIONS**



Outline Dimensions (millimeters)

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